



Article Fluid Dynamics Calculation in SF6 Circuit Breaker during Breaking as a Prerequisite for the Digital Twin Creation

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Abstract: The requirements to switching the capacities of SF6 circuit breakers submitted by Russian Grid companies are difficult to satisfy. The first limitation is related to material and financial costs in order to create a new requirement-satisfying switching device. The second limitation is dictated by the necessity of calculating complex physical processes in a circuit braker interrupter during faultcurrent making or breaking before creating a prototype. The latter task is reduced to the problem of simulating the processes of interaction between the switching arc and the SF6 gas flow. This paper deals with the solution of the problem both analytically by a special method and numerically by a numerical software package through the creation of a mathematical model of the interaction process. The switching arc is taken into account as a form of a temperature source, based on experimental data on measuring the temperature of the arc column. The key feature of the research is to use the finite element method based on a moving mesh-the Arbitrary Lagrangian Eulerian (ALE) method. Such a problem statement allows us to take the contact separation curve of the circuit breaker into account as the input data of the model. The calculations were carried out during fault-current breaking by a 110 kV SF6 dead-tank circuit breaker. The calculations of pressure and mass flow in the under-piston volume change, gas flow speed, and temperature depending on the contact separation are given. The proposed model of the switching arc was used to simulate the process of 25 kA symmetrical fault-current breaking and was compared with an experiment.

Keywords: computational fluid dynamics; SF6 circuit breaker; switching arc; moving mesh; ALE; arc quenching

MSC: 68T20

1. Introduction

The increase in power consumption in the Unified National Power Grid (UNPG) of Russia, along with the expansion of the technical and regulatory framework, quantitatively increases the used equipment or their replacements. One of the most important elements of the Electric Power System (EPS) ensuring its reliability is a such electrical device as a power circuit breaker (CB). Its main tasks are to interrupt short-circuit currents and isolate faulty parts of an EPS. However, conflicting requirements are simultaneously imposed on the CB. On the one hand, during contingencies—fault–current breaking or making—the CB must turn them off and provide an infinitely large resistance between the arcing contacts. On the other hand, under normal conditions, the operating currents through its contact system and its resistance must be infinitely small in order to avoid unnecessary losses of the power to be transmitted [1].

Thus, the following requirements are imposed on the CB:

- Low resistance in normal conditions (in the normally closed contact);
- High-voltage proof of external and internal insulation, which makes it possible to withstand lightning and switching overvoltage, as well as transient recovering voltage (TRV) after the arc is extinguished;



Citation: Popovtsev, V.V.; Khalyasmaa, A.I.; Patrakov, Y.V. Fluid Dynamics Calculation in SF6 Circuit Breaker during Breaking as a Prerequisite for the Digital Twin Creation. *Axioms* 2023, *12*, 623. https://doi.org/10.3390/axioms12070623

Academic Editor: Leonid Plotnikov

Received: 2 May 2023 Revised: 12 June 2023 Accepted: 19 June 2023 Published: 22 June 2023



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- The ability of both making and breaking the short-circuit currents —the CB must reliably extinguish the arc without its re-ignition;
- Ensuring fast transition from the closed to open position and vice versa, especially in automatic reclosing cycles.

In addition to the short-circuit currents, the CB must also provide the switching capacity of capacitive currents of unloaded overhead transmission lines, capacitor banks, and inductive currents of shunt reactors in accordance with Russian [2] and foreign [3–5] standards.

The principle of operation in most CBs as mechanical switching devices is based on actuating the operating mechanism—the drive. When the protective relay sends a command to the opening solenoid, the CB must operate within a very short period of time (the total opening time of gas CBs with a rated voltage of $U_{nom} = 110$ kV is 55 ± 5 ms).

When the contacts open, an electric arc occurs between them. At its core, an electric arc is an independent arc discharge, which is a low-temperature plasma channel characterized by a high current density and a low cathode voltage drop [6,7].

In CBs with a rated voltage of 110 kV and above, the electric arc is a burning highpressure arc in oil. It can also be compressed air or another gaseous dielectric with good arc-extinguishing properties, such as SF6 gas. The interruption of the current is carried out by cooling the arc plasma in such a way that the resulting electric arc disappears at the first occurring current zero after the contact separation. This process of cooling or extinguishing can be carried out in various ways, due to which the CBs are classified according to the type of arc quenching medium and the interrupter type. Currently, most CBs use compressed SF6 gas as an arc extinguishing medium—it has a high voltage proof (2.5 times higher than that of air) and a high heat transfer coefficient [8]. The arc extinguishing process itself lies in the fact that the arc under the high speed is blown by the cold high-pressure gas—puffer and self-blast technologies [9]. A modern interrupter is designed in such a way that the gas flow, cooling the arc, is supersonic (Mach numbers > 1) in order to level out the re-ignition of the arc in the next power frequency half-cycle when the current zero approaches [10].

Internationally, special scientific interest has been shown to SF6 as an arc-quenching medium in CBs. The main reason is that it is quite difficult to find a worthy alternative with the same arc-quenching characteristics so that the CB can provide high switching capacity [11]. Despite the active search for alternatives to SF6, due to its high global warming potential, an environmentally friendly and dielectrically equivalent medium has not been found. Thus, the leading manufacturers of high voltage CBs continue to use SF6.

In Russia, the interest in SF6 CBs is supported by the regulatory documents from the operator of the Russian power grids—the Federal Grid Company (FGC)—Rosseti (Regulation "On the Unified Technical Policy in the Integrated Power Grids"). It regulates the preferential use of SF6 CBs for the rated voltages of 110 kV and higher. In addition, according to the digital policy of the FGC–Rosseti (the concept of "Digital Transformation 2030"), one of the best methods to accurately control the operation of electrical equipment is to use an effective monitoring system.

Given the growth of short-circuit currents in power grids of 110 kV and above [12,13], there is a problem in increasing the CBs' switching capacities. One of the solutions is to optimize the design of the CBs or create new devices [14]. However, the development and design of high voltage switching devices is an expensive undertaking, as it requires numerous experimental tests, both on physical models and prototypes. In addition, there is a need to take into account the conflicting requirements for the CB interrupter in terms of its switching capacity, mechanical characteristics, electrical insulation level, etc., which also requires time and financial costs.

Thus, the calculation and modeling of internal processes occurring in the interrupter of SF6 CB is of particular scientific interest, both in the field of the operation of switching equipment and in the field of its design. Therefore, a course should be taken towards the development of approaches to modeling complex physical (gas-, thermo-, electrodynamic, and electrophysical) processes that occur in SF6 CB interrupters during fault–current making or breaking, with the possibility of verifying the resulting model—creating its digital twin.

2. Switching of SF6 Circuit Breakers

2.1. Interrupter Types in SF6 Circuit Breakers

Arc extinguishing in modern SF6 CBs occurs in the interrupter or the interruption chamber. These are special chambers in which the process of cooling the arc should be intensified, removing heat from it. Such a process is called blast, which is possible on exposure to the SF6 gas, flowing at the speed of sound, relative to the arc column. It is possible to organize effective blasting in SF6 CBs of 110 kV and higher in several ways, according to which the following designs of interrupters are distinguished [10,11,15–17]:

(1) Puffer type, single pressure: the blast is created by the means of a built-in compression device that creates excess pressure due to the drive's energy;

(2) Double pressure: equipped with a longitudinal blowing system, in which precompressed gas is supplied from a reservoir with a relatively high pressure of SF6 gas, which, after the arc is extinguished, undergoes a recompression process;

(3) Self-blast/Auto-puffer: blast occurs during the thermal expansion of SF6 gas using the energy of the arc itself to (partly) produce the pressure necessary to blast the arc;

(4) Magnetic arc rotation: interrupters with electromagnetic blast;

(4.1) The arc is extinguished by its rapid movement in a stationary SF6 gas under the influence of a radial magnetic field, created by arc itself;

(4.2) A longitudinal blowing system, in which the increase in pressure in the SF6 gas occurs when it is heated by an arc, rotating in a special chamber under the influence of a magnetic field.

Basically, in SF6 CBs with a rated voltage of 110 kV and above, the first two types of interrupter designs are used, along with the thermal expansion of SF6 gas under the action of a switching arc. For higher rated voltages (220 kV and above), or for higher switching capacity requirements for CBs with a voltage of 110 kV (rated breaking current of 50 kA and above), the operation of the interrupter can be further optimized. For example, by using the so-called double-motion principle, which consists of moving two arcing contacts in opposite directions. Another option is using a double-speed mechanism, which consists of dividing the mass of the moving contacts into two parts (upper and lower) and temporarily transferring part of the kinetic energy from the lower mass to the upper mass.

Single pressure SF6 circuit breaker operating.

In such designs, the overpressure that provides the gas flow is formed during the switching. Interrupter types are divided into a puffer type and a self-blast type.

Figure 1 shows a diagram of a double-blast, puffer-type interrupter. Inside the sealed insulating chamber filled with SF6 gas, the two contacts 1 and 2 are rigidly connected to each other. They are connected to the power drive mechanism through an insulating rod (is not shown in Figure 1). First, during the fault–current breaking, practically without discharges, the main contacts 1 and 3 open, and then the current passes into the arcing contact zone between contacts 2 and 5, where arc 4 burns. The insulating rod moves the entire moving system relative to the fixed piston 8, whereas, as it moves, SF6 gas is compressed in the working capacity of the cylinder cavity *B*—the under-piston volume. Thus, the principle of self-blast is implemented. Arc 4, which occurs between the arcing contacts 2 and 5, is drawn into the nozzles 6 and 7 by the flow of the compressed SF6 gas. A double-blast principle is provided, which intensively affects the arc shaft, which goes out in one of the current zeroes.



Figure 1. Puffer-type interrupter: 1—moving main contact; 2—moving arcing contact; 3—fixed main contact; 4—arc; 5—fixed arcing contact; 6—PTFE main nozzle; 7—PTFE auxiliary nozzle; 8—piston; A—above-piston volume; B—under-piston volume.

The improvement of SF6 CBs due to the increased requirements for switching capacity, on the one hand, is associated with an increase in the rated breaking current per one pole of the CB [16,17]. On the other hand, the goal is to reduce the power of the drive mechanism. However, these two methods contradict each other since a decrease in drive power causes the pressure drop at the moment of arc quenching, which means a decrease in breaking capacity. A fundamentally different way to increase the efficiency of the arc extinguishing of SF6 CBs during electric arc burning in the nozzle channel is possible when the nozzle is made of an insulating material—polytetra-fluoro-ethylene (PTFE). The impact of radiation energy on the inner surface of the insulating nozzle initiates an additional gas blast due to the ablation of the insulating walls, accompanied by the release of C2 and CF4. This leads to an increase in pressure in the contact gap and a consumption effect that limits the access of the arc-quenching medium to the contact gap at the maximum of the current to be broken. According to Figure 2a, the ablation of the insulating walls of the PTFE nozzle 2 occurs between the arcing contacts 3 and 4. This effect makes it possible to increase the gas pressure in chamber K not only due to the high temperature but also due to the additional mass flow from the gas-generating walls of the chamber [9,10].



Figure 2. Interrupter types: (a) Self-blast; (b) Single-blast; (c) Full double-blast.

Double-pressure SF6 circuit breakers.

In such designs, a pre-created pressure drop is formed during the shutdown process, which provides the gas flow. The arc-quenching devices of this type are divided into single-blast (Figure 2b) and double-blast (Figure 2c).

In the case of a longitudinal blast (Figure 2a), it is possible to obtain a better SF6 flow and the absence of the so-called "dead zone". In this case, the direction of gas movement coincides with the longitudinal axis of the interrupter and the axis of the switching arc. The arc that occurs between contacts 3 and 4 interacts with the longitudinal gas flow formed by the nozzle 2. The gas flow is provided by the pressure difference in the upper and lower

parts of the flow— p_0 and p_b , respectively. During the arc quenching, this difference is not constant. However, in the optimal case of arc quenching, it provides the supercritical gas outflow case ($p_0 > p_b$)—the critical gas mass flow rate for the longest possible period of time [9]. The main parameters of the interrupter that affect the formation of the gas flow are given in Section 3, according to [15,18].

In case of double blast (Figure 2b), the direction of the gas flow's movement, blowing over the arc, was the opposite. The arc 1 burned between the contacts 3 and 4 in the gas flow, formed by the two nozzles 2. These flows were formed by the channel made up of the tips of the contacts. The flow was first directed perpendicular to the arc and then rotated 90°. In this case, on the axis of the nozzle channel, where two radially directed jets met (at the point where the flow turns), a stagnation area, or a "dead zone", was formed, where the effect of the SF6 gas on the arc column was minimal. In this "dead zone", the residual arc column had an increased diameter, the decay processes of the residual arc shaft were slowed down, and a cloud of hot, conductive gas remained there for a long time, which negatively affected the extinguishing process [15].

CBs with double-pressure interrupter types are structurally complex. Firstly, they require an automatic compressor to recompress the gas from the low-pressure reservoir to the high-pressure one; secondly, the heating of the SF6 gas in the high-pressure reservoir [15,16] is required. The latter is related to the fact that SF6 liquefies at about 10 °C and at a pressure of about 1.6 MPa. Therefore, it is necessary to install heaters in the high-pressure reservoir to avoid SF6 liquefaction [8,16].

2.2. Models of Switching Arc Interaction with SF6 Gas Flow

All calculations of gas (fluid) dynamics processes are based on three well-known equations of fluid and gas mechanics:

- Continuity equation (law of conservation of mass);
- Equation of the second law (law of conservation of momentum);
- Energy equation (law of conservation of energy).

In order to describe the processes occurring in the arc within the framework of interaction with the SF6 gas flow, it is necessary to set a qualitative mathematical model.

Analytical models.

One of the first models was proposed by Cassie in 1939 [19]. It describes the process of arcing at high currents and is based on the assumption that the voltage of the arc column is constant. The model calculates an arc channel with a constant temperature, current density, and electric field strength. Changes in the conductivity of the arc are caused by the changes in the cross-section of the arc. The tension on the arc shaft is constant and does not depend on the arc current.

Mayer's model, proposed in 1943 [20], describes the process of arcing at currents, close to zero, and is based on the assumption that the power removed from the arc is unchanged:

$$\frac{1}{g} + \frac{dg}{dt} = \frac{d\ln g}{dt} = \frac{1}{\tau} \left(\frac{ui}{P} - 1 \right) \tag{1}$$

where *g* is arc conductivity; *t* is arcing time; *u* is arc voltage; and *i* is arc current.

The arc is represented as a non-linear resistance in an equivalent circuit. The main idea of the model is that only convection causes power losses, i.e., the temperature in the arc is not constant. This means that the cross-sectional area of the arc is proportional to the current and that the arc voltage is constant. Furthermore, it is believed that power losses are caused by thermal conduction at low currents. This means that the conductivity strongly depends on temperature but does not depend on the cross-sectional area of the arc.

The Cassie and Mayer models have modifications—hybrid models that allow a more accurate description of the breaking arc, for example, Brown's model [21], which he subsequently applied to analyze the breaking process in the post-zero arcing period, characterized by energy balance [22]. Undoubtedly, the above models are useful but have limited appli-

cations, as they are based on ordinary differential equations. In other words, they cannot be used to study in detail the physical processes during arcing, i.e., evaluate the Interaction with the SF6 gas flow since it is described by the equations of gas dynamics, which are differential equations in partial derivatives [23].

Modified arc models.

More promising arc models [24–29] have limitations in their uses, described in [23]. They are based on the continuity, momentum, and energy equations of gas dynamics and Ohm's Law.

Continuity equation:

$$\frac{\partial \rho}{\partial t} + \frac{\partial (\rho \vartheta_{\mathbf{z}})}{\partial \mathbf{z}} + \frac{1}{\mathbf{r}} \cdot \frac{\partial (\mathbf{r} \rho \vartheta_{\mathbf{r}})}{\partial \mathbf{r}} = 0$$
(2)

Axial momentum equation:

$$\rho \frac{\partial \vartheta_{\mathbf{z}}}{\partial t} + \rho \vartheta_{\mathbf{z}} \frac{\partial \vartheta_{\mathbf{z}}}{\partial \mathbf{z}} + \rho \vartheta_{\mathbf{r}} \frac{\partial \vartheta_{\mathbf{z}}}{\partial \mathbf{r}} = -\frac{\partial p}{\partial \mathbf{z}} + \frac{1}{\mathbf{r}} \cdot \frac{\partial}{\partial \mathbf{r}} \left[(\eta + \eta_t) \cdot \mathbf{r} \frac{\partial \vartheta_{\mathbf{z}}}{\partial \mathbf{r}} \right]$$
(3)

Energy equation:

$$\rho \frac{\partial h_0}{\partial t} + \rho \vartheta_{\mathbf{z}} \frac{\partial h_0}{\partial \mathbf{z}} + \rho \vartheta_{\mathbf{r}} \frac{\partial h_0}{\partial \mathbf{r}} = \sigma \mathbf{E}^2 - U + \frac{1}{\mathbf{r}} \cdot \frac{\partial}{\partial \mathbf{r}} \left[(k + k_t) \cdot \mathbf{r} \frac{\partial T}{\partial \mathbf{r}} \right]$$
(4)

Ohm's law:

$$I = \mathbf{E} \int_{0}^{t_1} 2\pi \mathbf{r} \boldsymbol{\sigma} \cdot d\mathbf{r}$$
(5)

In the equations above, ρ is gas density; ϑ_z is axial velocity; ϑ_r is radial velocity; p is gas pressure; η is molecular viscosity; η_t is turbulent viscosity; σ is electrical conductivity; E is the voltage gradient; U is the net emission coefficient; k is thermal conductivity; k_t is turbulent thermal conductivity; and h_0 is total enthalpy.

Generally, the axial symmetric problem statement means that Equations (2)–(5) are integrated over the radius within limits from a to b, and equations from [23] are obtained.

Based on the above integral equations of the arc, it is possible to adequately describe the interaction of the arc with the blown SF6 gas flow. In other words, in contrast to the problems of magnetohydrodynamics, in which the movement of a conducting gas in an electromagnetic field manifests itself in two effects—the Lorentz force and the Joule heat release—the calculation of processes in arc extinguishing devices is reduced to a gas-dynamic problem in the presence of only one electromagnetic effect—Joule heat release. This principle of arc modeling has been described in studies [14,30–33].

To calculate gas dynamics, you can also use the technique from [15,18], which is based on the analytical determination of the parameters of the piston device and its speed, in order to obtain a pressure drop that provides a supercritical gas flow mode. To determine the expiration mode, the pressure ratio under the piston p_0/p_i is considered, where *i* is the calculation step, and p_0 is the initial pressure in the under-piston volume. In addition to the serious assumptions used, the main disadvantage of this technique is that it does not take into account the interaction of the gas flow with the arc.

Experimental KEMA model.

The model is based on 79 short-circuit tests of CBs with a rated voltage range of 123–550 kV in test center "KEMA" (now "CESI") [34–37]. The model accuracy of the interruption prediction (prediction of post-current-zero events from pre-current-zero information). The arc parameters for each arc model are extracted by an optimization approach in a defined time interval to minimize the difference between the measured and simulated voltage waveforms.

At its core, this experimental model is a modified Mayer model, assembled from three submodels: $dg_i = -\frac{1}{2} \cdot \alpha^{\lambda_i} \cdot \mu^2 - \frac{1}{2} \alpha_i$

$$\frac{{}^{4}\!S_{i}}{dt} = \frac{1}{P_{i}\tau_{i}} \cdot g_{i}^{A_{i}} \cdot u_{i}^{2} - \frac{1}{\tau_{i}}g_{i},
\text{for } i = 1, 2, 3$$
(6)

By solving each equation:

$$\frac{1}{g} = \frac{1}{g_1} + \frac{1}{g_2} + \frac{1}{g_3} \tag{7}$$

The model has six arc parameters: three time constants (τ_i) and three cooling power constants (Pi) [36], $\lambda_1 = 1.4$, $\lambda_2 = 1.9$, and $\lambda_3 = 2.0$. The model consists of three differential equations of g = i/u, the same ones as the earlier model, each representing different time intervals of the interruption processes (Figure 3).

$$i \longrightarrow \boxed{\frac{dg_1}{dt} = \frac{1}{P_1 \tau_1} g_1^{1.4} u_1^2 - \frac{1}{\tau_1} g_1} \longrightarrow \boxed{\frac{dg_2}{dt} = \frac{1}{P_2 \tau_2} g_2^{1.9} u_2^2 - \frac{1}{\tau_2} g_2} \longrightarrow \boxed{\frac{dg_3}{dt} = \frac{1}{P_3 \tau_3} g_3^2 u_3^2 - \frac{1}{\tau_3} g_3} \longrightarrow \underbrace{u_1 \longrightarrow u_2} \longrightarrow \underbrace{u_2 \longrightarrow u_3} \longrightarrow \underbrace{u_3 \longrightarrow u_3}$$

Figure 3. The tree-part arc model of three arcs in series.

Parameters of the KEMA model are represented in Table 1.

Table 1. KEMA model parameter	rs.
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Parameter Description	Parameter		
	Symbol	Formula	
Arc parameters	(varying from test to test)		
Time constant	$ au_1$	$\frac{k_t}{l_a - l_T}$	
Cooling power constants	B_1	-	
	<i>B</i> ₂	-	
Parameters	related to CB design		
Distance between arcing contacts	l_a	-	
Empirical constant (depends on tested CB and for conditions of the short-line fault)	l_t, k_t	According to [3]	
Time constants	$ au_2$	$\frac{\tau_1}{k_1}$	
	τ ₃	$\frac{\tau_2}{k_2}$	
Constants representing the breaker design	k_1	According to [37]	
	k_2	According to [37]	
	k_3	According to [37]	
Cooling power	P_1	$B_1 \cdot g_1^{0.6}$	
	P2	$B_2 \cdot g_2^{0.1}$	
	<i>P</i> ₃	$\frac{B_2}{k_3}$	

Magnetohydrodynamic model.

Fluid models describe plasma in terms of smoothed quantities, such as density and average velocity around each position. The fluid model in the magnetohydrodynamic approach considers plasma as a single fluid, as described by the systems of the Maxwell and Navier–Stokes equations [7]. A more general description is a two-fluid plasma, where ions and electrons are described separately. Fluid models are often accurate when the collision probability is high enough to keep the plasma velocity distribution close to the Maxwell–Boltzmann distribution [38]. Fluid models typically describe plasma in terms

of a single flow at a certain temperature in each spatial location. They cannot capture high-velocity spatial structures, such as beams or double layers, nor resolve particle wave effects [39,40].

The gas dynamics and electrodynamics of the electric arc, in addition to the equations of continuity (2), motion (3), energy (4), and Ohm's law (5), must be supplemented with Maxwell's equations and material relations:

$$\frac{\mathbf{E}_{\mathbf{r}}}{\partial \mathbf{z}} - \frac{\partial \mathbf{E}_{\mathbf{z}}}{\partial \mathbf{r}} = 0 \tag{8}$$

$$\frac{1}{\mathbf{r}}\frac{\partial \mathbf{r}\mathbf{H}_{\boldsymbol{\varphi}}}{\partial \mathbf{r}} = j_z \tag{9}$$

$$\frac{\partial \mathbf{H}_{\boldsymbol{\varphi}}}{\partial \mathbf{z}} = j_r \tag{10}$$

where according to Ohm's law:

$$j_z = \sigma \mathbf{E}_{\mathbf{z}}; \ j_r = \sigma \mathbf{E}_{\mathbf{r}}; \tag{11}$$

$$\mathbf{B} = \mu_0 \mathbf{H}; \mathbf{D} = \varepsilon_0 \mathbf{E}; \tag{12}$$

where μ_0 is magnetic permeability, and ε_0 is dielectric permittivity.

In general, μ_0 and ε_0 are tensors.

The above equations are supplemented by dependencies:

$$\rho = \rho(T, p); \ \sigma = \sigma(T, p); \ \lambda = \lambda(T, p); \ \eta = \eta(T, p); \ c_p = c_p(T, p); h = h(T, p); \ \psi = \psi(T, p)$$
(13)

Boundary conditions in this case (under axisymmetric conditions—cylindrical coordinates):

$$\mathbf{r} = 0, z > 0; \ \boldsymbol{\vartheta}_{\mathbf{r}} = 0; \ \mathbf{H}_{\boldsymbol{\varphi}} = 0; \ \frac{\partial \boldsymbol{\vartheta}_{\mathbf{z}}}{\partial \mathbf{r}} = 0; \ \frac{\partial T}{\partial \mathbf{r}} = 0; \ \frac{\partial \mathbf{E}_{\mathbf{z}}}{\partial \mathbf{r}} = 0; \ \boldsymbol{\omega} = 0;$$
 (14)

where $\mathbf{E}_{\mathbf{r}}$ and $\mathbf{E}_{\mathbf{z}}$ are radial and axial voltage gradients, respectively; $\mathbf{H}_{\boldsymbol{\varphi}}$ is azimuthal magnetic field strength; j_r and j_z are radial and axial electric current density, respectively; ρ is the gas density; p is pressure; λ is thermal conductivity; c_p is specific heat capacity at constant pressure; h is specific enthalpy; and ψ is emissivity.

The equations written above can be used to describe an arc discharge in a gas. However, from the side of low currents, the boundary of the region under study is determined, as a rule, by the fulfillment of the condition of local thermodynamic equilibrium. When high currents are turned off, the limitation occurs due to the influence of reabsorption from radiation.

Hydrokinetic model.

The uniqueness of this model lies in the fact that it has a huge advantage over the magnetohydrodynamic approach. The reason is that it is possible to calculate the processes of restoration of the dielectric strength after the extinction of the arc within the framework of hydrokinetic modeling. The model is characterized by four calculation stages [41,42]:

(1) The arc plasma properties (in insulation media as SF6) for basic input data are needed, including thermodynamic properties, transport coefficients, and radiation coefficients, which are being calculated according to the Chapman–Enskog theory under local thermodynamic equilibrium (LTE) [40];

(2) A 1D hydrokinetic model is being used for the arc decaying process description (assuming cylindrical symmetry);

(3) Based on the 1D modeling results, the average radial temperature, the arc conductance, and the average critical electric field strength are calculated the three recovery stages: thermal recovery rate, the predielectric recovery rate, and the postdielectric; (4) The postdielectric recovery stage is being calculated relying on the Boltzmann equation [43], which describes the electron transport behaviors during the last phase of extinguishing arcs.

In this model, the arc was assumed to be wall stabilized in a cylinder with 1D geometry. The governing equations describing this 1D arc were written as follows:

Continuity equation:

$$\frac{\partial \rho}{\partial t} + \frac{1}{\mathbf{r}} \cdot \frac{\partial (\mathbf{r} \rho \vartheta_{\mathbf{r}})}{\partial \mathbf{r}} = 0$$
(15)

Energy equation:

$$\rho c_p \left(\frac{\partial T}{\partial t} + \vartheta_{\mathbf{r}} \frac{\partial T}{\partial \mathbf{r}} \right) = \sigma \frac{i^2}{g^2} - E_{rad} + \frac{1}{\mathbf{r}} \cdot \frac{\partial}{\partial \mathbf{r}} \left(k \mathbf{r} \frac{\partial T}{\partial \mathbf{r}} \right)$$
(16)

where **r** is radial distance; *q* is mass density; $\vartheta_{\mathbf{r}}$ is radial component of the velocity; c_p is specific heat at constant pressure; *T* is temperature; σ is electrical conductivity; *j* the thermal conductivity; *i* is current; *g* is arc conductance; and E_{rad} is radiation energy loss (net emission coefficient *U* in (4)).

One of the significant disadvantages of this model is the huge computing power. However, if the problem statement is made in 1D, then the calculation time will be significantly reduced.

Kinetic model.

Kinetic models describe the particle velocity distribution function at each point in the plasma and, therefore, should not assume a Maxwell–Boltzmann distribution. A kinetic description is often necessary for a collision-less plasma [44,45]. Such a plasma can be considered as the one in which the density is low enough and/or in which the temperature is high enough so that collisions can be neglected, due to the fact that the characteristic times are shorter than the particle collision time [39,44].

There are two general approaches to the kinetic description of plasma. One is based on the representation of the smoothed distribution function on the grid in terms of velocity and its position (distribution function of particles in coordinates and momenta):

$$f = f(t, \mathbf{r}, \mathbf{p}) \tag{17}$$

where **r** is particle center of mass coordinates, and **p** is the impulses of the center of mass of particles.

Function (17) in the LTE state has the form of a Maxwellian distribution and is generally found from the Boltzmann equation:

$$\frac{\partial f}{\partial t} + \vartheta \frac{\partial f}{\partial \mathbf{r}} + F \frac{\partial f}{\partial \rho} = C(f)$$
 (18)

wherein:

$$\mathbf{F} = e\mathbf{E} + \left(\frac{e}{c}\right)[\boldsymbol{\vartheta}\mathbf{B}] \tag{19}$$

where **F**—an external force acting on a charged particle; and C(f)—taking into account mutual collisions of particles.

When considering fast motions of particles, collisions can often be neglected, assuming $C(f) \approx 0$. Then, the kinetic equation is called the collision-less Vlasov equation with self-consistent fields **E** and **B** (they are themselves determined by the motion of charged particles).

The Vlasov equation can be used to describe the dynamics of a system of charged particles interacting with an electromagnetic field. In magnetized plasma, the hydrokinetic approach can significantly reduce the computational cost of a fully kinetic simulation [40,45,46].

Another method, known as the particle-in-cell method, incorporates kinetic information by following the trajectories of a large number of individual particles. In fact, this method is used to solve nonstationary problems of magnetohydrodynamics [44]. Kinetic models tend to be more computationally expensive than fluid models.

Analyzing the works devoted to the modeling of gas-dynamic processes in SF6 CBs during their switching, the main focus was on solving the following scientific and practical problems:

1. An exploration of the electrophysical impact of the arc energy on the nozzle ablation process in order to both create advanced systems for monitoring the residual switching life and to study the effect of an auto-puffer to increase the switching capacity [47–49];

The optimization of the design of elements of the arc quenching device, the nozzle section in particular, to increase the flow rate of SF6 and the switching capacity of the CBs [50,51];

An evaluation of the residual conductivity of the arc stem to study its effect on the power proof of the contact gap after its extinction [52,53];

The main difficulty in the vast majority of studies has been related to the implementation of a suitable mathematical model of heat and mass transfer of a cold SF6 flow and a non-isothermal plasma channel. Some generalizations of works devoted to the construction of mathematical models of the interaction of an SF6 flow with an opening arc in SF6 CBs are given in Table 2.

₽	Ref.	Problem under Study	The Model of Arc Interaction with SF6 Flow	Computational Numerical Model	
1	[32]	Predicting arc extinction by simulating outgassing with nozzle ablation	Conservation equations, Joule heating, and radiation transfer	A two-dimensional axisymmetric	
2	[33]	Exploration of the arc extinguishing process, when the capacitive current is turned off by a self-generating switch	Conservation equations, Joule heating, radiation transfer	A two-dimensional axisymmetric	
3	[47]	Exploration of the nozzle ablation process for breaking capacity	Conservation equations, radiation transfer	A two-dimensional planar	
4	[54]	Elimination of an impulse wave in front of a stationary arcing contact inside the nozzle, causing a decrease in the flow rate of SF6 gas in the nozzle	Conservation equations	A two-dimensional planar	
5	[55]	Arc re-ignition prediction	Conservation equations, Joule heating and radiation transfer	A two-dimensional axisymmetric	
6	[56]	Influence of impurities, arising in the process of nozzle ablation on the process of arc quenching	Conservation equations	A two-dimensional planar	
7	[57]	The reconstruction of a digital model of an arc in cylindrical nozzles	Conservation equations	A two-dimensional planar	
8	[58]	Exploration of the influence of the aperiodic component of the tripping current on the process of arcing	Magnetohydrodynamic: conservation equations, Maxwell's equations	A two-dimensional axisymmetric	
9	[59]	Creation of a software package for modeling arc extinguishing processes	Conservation equations, Joule heating	A two-dimensional planar	
10	[60]	Exploration of the process of arc extinguishing by a self-blast CB, taking into account the ablation of the nozzle	Conservation equations	A two-dimensional axisymmetric	
11	[61]	Investigation of the process of arc extinguishing by a self-generated switch, taking into account the ablation of the nozzle	Conservation equations, radiation transfer	A two-dimensional axisymmetric	
12	[62]	Exploration of the arc extinguishing process in a supersonic nozzle	Conservation equations	A two-dimensional axisymmetric	
13	[63]	Improved accuracy at low breaking currents (wire arc).	Magnetohydrodynamic: conservation equations, Maxwell's equations	3D	

Table 2. Studies devoted to interaction of SF6 gas flow with the arc.

2.3. Methods for Calculating the Processes of Interaction between Arc and the SF6 Flow

Analytical methods.

Considering that the flow of SF6 gas is described by the system of Navier–Stokes equations, it was worth noting that analytical integration was possible only in a limited number of cases. Furthermore, most methods for calculating the equations of fluid dynamics today are reduced to numerical ones. In magnetohydrodynamics, when the system of Maxwell equations is added to the Navier–Stokes equations, the situation only worsens in the framework of precisely finding the analytical solution. The fact is that the significant multidimensionality of the processes under study, especially in the plasma channel of the shutdown arc blown by a nonisothermal flow, as well as the presence of various types of weak and strong discontinuities, made it difficult to use numerical analysis methods [64]. Therefore, analytical studies aimed at describing the features associated with the nonlinear and multidimensional nature of plasma motions based on exact solutions are topical. A universal method that allows for analytically solving nonlinear equations that describe magnetohydrodynamics is the method of group analysis of differential equations [65]. Group theory methods were later applied to problems in fluid mechanics [66].

Numerical methods.

Numerical modeling of the processes occurring in the tripping arc mainly prevailed over other methods due to the growth of computing power. First of all, this referred to the calculations of gas-dynamic fields, current density fields, and electromagnetic forces. This, in turn, led to the so-called pinch effect and plasma acceleration and also to a more correct description of the radiation transfer in the arc column. The main numerical methods of calculation, as applied to the problems of fluid mechanics, including magnetohydrodynamics, were the finite element method and the finite volume method [67].

Non-Numerical methods.

For the calculation of physical processes that have uncertainties in the formulation of the problem or its solution (in particular, only some particular solution is found by numerically solving the system of Navier–Stokes equations), polynomial chaos methods are gaining popularity. The essence of the method is to represent random processes on a stochastic polynomial space in the form of Hermite polynomials [68].

In this study, the solution of the problem of gas dynamics—the determination of the pressure and flow rate of SF6 gas—was implemented analytically (according to the method [15,18]) and numerically (by the finite element method, in the numerical simulation software COMSOL Multiphysics 6.0).

3. Analytical Calculation of SF6 Circuit Breaker Breaking

Analytical calculation was carried out according to the method specified in [15,18]. Based on the methodology, the following main assumptions were made:

- (1) There was no supply and removal of heat during the outflow of gas (adiabatic process);
- (2) The process of gas outflow had a steady character;
- (3) There were no friction losses;
- (4) The gas was considered ideal;

All the main ratios necessary for the calculation were taken from [15]. The block diagram of the analytical calculation could be represented as follows—Figure 4:



Figure 4. Block diagram of the analytical method: S_p —piston cross section; l—full contact separation; p_0 —pressure in the under-piston volume during the calculation; k_a —adiabatic index for SF6 gas; V_0 —initial volume of gas under the piston; M0—initial mass of SF6 gas under the piston; γ_0 —initial density of SF6 gas in the under-piston volume; γ_i —SF6 density in the under-piston volume during the calculation; $Y = p_0 / p_i$ —relative backpressure.

The main purpose of the calculation was to determine the parameters of the piston device and its speed in order to obtain a pressure drop that provided a supercritical gas outflow regime.

3.1. SF6 Circuit Breaker under Study

Dead-tank SF6 CB for rated voltage 110 kV(RU)/126 kV(EU)—Figure 5 was chosen as the research object for the calculation of gas-dynamic processes.



Figure 5. A 110 kV dead-tank SF6 CB: 1—moving main contact; 2—moving arcing contact; 3—fixed main contact; 4—arc; 5—fixed arcing contact; 6—PTFE main nozzle; 7—PTFE auxiliary nozzle; 8—piston; 9—tube of fixed main contact; 10—tube of moving main contact; 11—valve in the piston; 12—valve closed when contacts are opened; A—above-piston volume; B—under-piston volume.

The current flow path in the arc extinguisher of the selected CB is conceptually shown in Figure 1: in the closed state, most of the current flows from the live part of the bushing through the tulip contact (is not shown) to the tubes of both the fixed and moving main contacts 9 and 10, passing to the main contacts 1 and 3. When the main contacts open, the current also flows through the tubes 9 and 10 but passes through the conductive parts into the arcing contacts 5 and 7. Arc extinguishing in the interrupter of this CB is based on combining self-blast and puffer-type principles. The interrupter had PTFE nozzles, providing an additional gas blast when the arc is burning.

3.2. Computational Model of the Circuit Breaker under Study

To calculate the gas dynamics, the following data were required: the full contact separation, the cross section of the piston, as well as the characteristics of the dependence of the cross section of the SF6 gas outlet on the contact separation S = f(l). All these parameters were taken from the approximate geometric dimensions of the tank CB under study:

- Full contact separation $L_{max} = 120$ mm;
- The contact separation before blast start is $L_{ext} = 18$ mm;
- Piston cross section $S_p = 8.953 \text{ mm}^2$;
- Ambient medium temperature $\vartheta = 40 \circ C = 313$ K;
- Pressure inside CB $p_0 = 0.42$ Mpa;
- The flow coefficient μ at all stages of the outflow was assumed to be 0.9 (the outflow coefficient, which took into account the decrease in the actual cross section of the hole due to the compression of the jet in it);
- Adiabatic exponent for SF6 gas $k_a = 1.086$;
- We set the discretization step of the calculation; for this, we divided the entire piston stroke into 20 identical sections: n = 20. The contact separation in each section would be:

$$\Delta l = \frac{L_{max}}{n} = \frac{120}{20} = 6 \text{ mm}$$
(20)

Figure 6a,b shows the dependence of the SF6 outlet cross section on the piston stroke S = f(l) and the dependence of the piston speed on the contact separation V = f(l) for the selected research object—110 kV SF6 dead-tank CB. The data were taken from the instruction manual of the manufacturer of this CB. Figure 7 shows the contact travel curve, which was taken from [14]. This curve was necessary in numerical implementation when solving a gas-dynamic problem with a moving grid. The total shutdown time was taken as equal to 55 ms.



Figure 6. (a) Dependence of the SF6 outlet cross section on the contact separation; (b) Dependence of the piston velocity on the contact separation.



Figure 7. Dependence of moving contact separation on time.

3.3. Calculation Results

The results of the analytical calculation were the pressure changes in the under-piston volume and the mass flow rate per second depending on the contact separation. Table 3 presents the results of calculating each step by the analytical method according to the method [15,18]: pressure in the under-piston volume pi; gas mass flow *G_i*; mass loss of SF6 during blast; ΔM_i ; mass of SF6 gas in the under-piston volume M_i ; and relative backpressure Y_i . The piston speed vav was taken from Figure 6b. Parameter $\Psi(Y) = \sqrt{\frac{2k_a}{k_a-1}\left(Y^{\frac{2}{k_a}}-Y^{\frac{k_a+1}{k_a}}\right)}$ characterized the outflow of gas from the volume above the piston

for each *i* calculation step.

l, mm	V _i , mm ³	p _i , MPa	Yi	Ψ_{i}	v _{avg.i} , m/s	G _i , kg/s	$\Delta M_{i\prime} \ kg \cdot 10^3$	M_i , kg $\cdot 10^3$
6	1.209	0.420	1.000	0	0	0	0	29.674
12	1.155	0.441	0.975	0.218	0.40	0	0	29.674
18	1.101	0.465	0.938	0.336	1.15	0	0	29.674
24	1.048	0.491	0.895	0.424	1.80	0.114	0.380	29.294
30	0.994	0.512	0.856	0.482	2.40	0.271	0.676	28.617
36	0.940	0.530	0.823	0.520	3.10	0.455	0.881	27.736
42	0.886	0.547	0.795	0.546	4.15	0.494	0.715	27.022
48	0.833	0.569	0.766	0.569	5.04	0.533	0.635	26.387
54	0.779	0.596	0.734	0.588	5.28	0.574	0.652	25.735
60	0.725	0.627	0.701	0.604	5.28	0.863	0.981	24.755
66	0.671	0.653	0.671	0.614	5.28	1.177	1.337	23.417
72	0.618	0.673	0.646	0.620	5.28	1.231	1.399	22.019
78	0.564	0.695	0.625	0.623	5.28	1.279	1.453	20.565
84	0.510	0.719	0.604	0.625	5.28	1.326	1.507	19.058
90	0.457	0.747	0.582	0.625	5.28	1.374	1.561	17.497
96	0.403	0.780	0.559	0.625	5.28	1.427	1.622	15.875
102	0.349	0.820	0.535	0.625	5.14	1.491	1.740	14.135
108	0.295	0.867	0.508	0.625	4.40	1.564	2.133	12.002
114	0.242	0.902	0.486	0.625	3.30	2.360	4.291	7.711
120	0.188	0.733	0.526	0.625	1.95	2.188	6.732	0.979

Table 3. Gas dynamics parameters obtained analytically.

Pressure changes in the under-piston volume and mass flow rates depending on the contact separation, plotted according to Table 3 data, are shown in the calculation results in comparison with the numerical calculations.

4. Numerical Calculation of SF6 Gas Circuit Breaker Switching

The analysis was carried out in COMSOL Multiphysics 6.0 software (Academic version) using the finite element method with a moving mesh (ALE).

The Navier–Stokes equations describe nonstationary flows and express momentum balances and mass conservations for a viscid and compressible fluid in the following form [69,70]:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0 \tag{21}$$

where ρ —density; **u**—flow velocity m/s; and *t*—time, s.

$$\rho \frac{\partial \mathbf{u}}{\partial t} + \rho \mathbf{u} \cdot \nabla \mathbf{u} = -\nabla \mathbf{p} + \nabla \cdot \left(\mu \left(\nabla \mathbf{u} + (\nabla \mathbf{u})^T \right) - \frac{2}{3} \mu (\nabla \cdot \mathbf{u}) \mathbf{I} \right) + \mathbf{F}$$
(22)

where *p*—pressure, Pa; **F**—volume force vector N/m^3 ; and **I**—turbulence intensity.

The Reynolds number is one of the important criteria in the analysis. The Reynolds number (*Re*) is non-dimensional and describes the ratio between inertial forces and those of viscous friction in viscid fluids and gases.

When the Reynolds number exceeded a boundary value, the precise analytical solution for the dimensional flow or flat plate flow became chaotic, which marked the emergence of the turbulent flow. The Navier–Stokes equations were highly sensitive to changes in coefficient values in the turbulent flow conditions [69,70].

Values of the Reynolds number varied within $(0.4 - 10) \cdot 10^6$ depending on SF6 gas pressure in interruption chamber and geometrical properties of SF6 gas concurrent flow, which blew round the arc [71,72].

There were plenty of various models for turbulent flow analysis. The main idea behind such a model boiled down to the assumption about the existence of average flow velocity and the average deviation from flow velocity. The models that are listed below were used in various engineering applications with different accuracy requirements. Almost all of them have been implemented in modern software for fluid dynamics analysis.

The main models are listed in the order of increasing complexity [67,70]:

- The Boussinessq model;
- The Spallart–Allmaras model;
- The $k \varepsilon$ model;
- The $k \omega$ model;
- The Reynolds stress model;
- The direct numerical simulation (DNS);
- The large eddy simulation.

The $k - \varepsilon$ model was chosen to be used in numerical analysis, with the turbulent flow of gas being taken into account. This model is mainly used in nozzle arc simulations [73,74]. Furthermore, the $k - \omega$ model was used to simulate CB interruption without arc (no-load mode) due to existence of a boundary layer, which is described below. This model could be efficiently applied in the wall turbulence analysis without additional special functions.

4.1. The $k - \varepsilon$ Model of Turbulent Flow

The turbulence viscosity ratio μ_T for the $k - \varepsilon$ model is:

$$\mu_T = \rho C_\mu \frac{k^2}{\varepsilon} \tag{23}$$

where ε is velocity of turbulent dissipation, m²/s³; C_{μ} is coefficient of the $k - \varepsilon$ model; and k is turbulent kinetic energy, m²/s².

The resultant equations of the $k - \varepsilon$ model are the following [75]: The convection–diffusion equation for turbulent kinetic energy *k*:

$$\rho \frac{\partial k}{\partial t} + \rho(\mathbf{u} \cdot \nabla)k = \nabla \cdot \left[\left(\mu + \frac{\mu_T}{\sigma_k} \right) \nabla \mathbf{k} \right] + \mathbf{P}_k - \rho \varepsilon$$
(24)

The convection–diffusion equation for the dissipation ε :

$$\rho \frac{\partial \varepsilon}{\partial t} + \rho(\mathbf{u} \cdot \nabla)\varepsilon = \nabla \cdot \left[\left(\mu + \frac{\mu_T}{\sigma_{\varepsilon}} \right) \nabla \varepsilon \right] + C_{\varepsilon 1} \frac{\varepsilon}{k} P_k - C_{\varepsilon 2} \rho \frac{\varepsilon^2}{k}$$
(25)

where ε , $C_{\varepsilon 1}$, and $C_{\varepsilon 2}$ are model parameters, and their values are listed in Table 4.

 Table 4. Computational model parameters for numerical calculation.

Description	Parameter				
	Designation	Value			
Pressure inside the interrupter	р	0.42 MPa			
Initial gas flow velocity	и	0 m/s			
Ambient temperature	Т	313 K			
Von Karman constant	k_v	0.41			
Parameters of	f k-e turbulence model				
_	$C_{arepsilon1}$	1.44			
_	$C_{\epsilon 2}$	1.92			
_	C_{μ}	0.09			
Turbulent kinetic energy	σ_k	1			
Turbulent dissipation rate	$\sigma_{arepsilon}$	1.3			
Constant parameters of k-w turbulence model					
_	α	0.12			
_	eta_0	0.072			
_	$egin{array}{c} eta_0^* \end{array}$	0.09			
Turbulent kinetic energy	σ_k^*	0.5			
Specific turbulent dissipation rate	σ_{ω}	0.5			

The component of deformation rate P_k is:

$$\mathbf{P}_{k} = \mu_{T} \left[\nabla \mathbf{u} : \left(\nabla \mathbf{u} + \left(\nabla \mathbf{u} \right)^{T} \right) - \frac{2}{3} \left(\nabla \cdot \mathbf{u} \right)^{2} \right] - \frac{2}{3} \rho k \nabla \cdot \mathbf{u}$$
(26)

where «:» means tensor convolution.

4.2. The $k - \omega$ Model of Turbulent Flow

The turbulence viscosity ratio μ_T for the $k - \omega$ model is:

$$\mu_T = \rho \frac{k}{\omega} \tag{27}$$

where ω is relative dissipation rate, Hz.

The resultant equations of the $k - \omega$ model are the following [75]:

The convection–diffusion equation for turbulent kinetic energy *k*:

$$\rho \frac{\partial k}{\partial t} + \rho(\mathbf{u} \cdot \nabla)k = \nabla \cdot [(\mu + \sigma_k^* \mu_T) \nabla \mathbf{k}] + \mathbf{P}_k - \rho \beta^* k \omega$$
(28)

The equation of relative dissipation rate ω :

$$\rho \frac{\partial \omega}{\partial t} + \rho(\mathbf{u} \cdot \nabla)\omega = \nabla \cdot \left[(\mu + \sigma_{\omega} \mu_T) \nabla \omega \right] + \alpha \frac{\omega}{k} \mathbf{P}_k - \rho \beta \omega^2$$
(29)

where α ; $\beta = \beta_0 f_{\beta}$; $\beta^* = \beta_0^* f_{\beta}$; σ_{ω} ; σ_k^* ; β_0 ; $f_{\beta} = \frac{1+70\chi_{\omega}}{1+80\chi_{\omega}}$; and $\chi_{\omega} = \left|\frac{\Omega_{ij}\Omega_{jk}S_{ki}}{(\beta_0^*\omega)^3}\right|$ —are model parameters, and their values are listed in Table 4.

$$\Omega_{ij} = \frac{1}{2} \left(\frac{\partial \bar{u}_i}{\partial x_j} - \frac{\partial \bar{u}_j}{\partial x_i} \right)$$
 is the tensor of average rotation rate;

$$S_{ij} = \frac{1}{2} \left(\frac{\partial \bar{u}_i}{\partial x_j} - \frac{\partial \bar{u}_j}{\partial x_i} \right)$$
 is the tensor of average deformation rate:

Furthermore, the wall boundary conditions were formed for the $k - \omega$ model [75].

4.3. The Computational Model of the Object under Study

The model of interruption chamber of a dead-tank SF6 gas CB. The dimensions of the model were approximately the same as those of the real CB. Elements of the chamber, which were not utilized in gas dynamics analysis, were not taken into account (Figure 8).



Figure 8. Computational model for numerical calculation (axial symmetry): 1—fixed arcing contact; 2—moving arcing contact; 3—auxiliary PTFE nozzle; 4—main PTFE nozzle; A—above-piston volume; B—under-piston volume.

Parameters of the turbulence models k-e and k-w for numerical analysis can be found in Table 4.

The grid of the numerical model was triangular (the triangulation method). The rectangular domain was used at the boundary layer. Grid parameters are shown in Table 5. The calculation was carried out using the finite element method in combination with the arbitrary Lagrangian Eulerian method (ALE). The latter was used to solve gas dynamics equations in the third coordinate system, thus called mesh frame.

Table 5. Mesh and time dependent solver parameters.

Number of Elements	Vertex Elements	Edge Elements	Average Element Quality	Automatic Remeshing	Relative Tolerance	Tolerance Factor	Termination Technique	Max Iterations
4737	92	1090	0.4474	0.08	0.1	1	Tolerance	20

This study was distinguished by taking into account the movement of the mobile parts of the object under study in a gas dynamics simulation. The module moving mesh was used to simulate the movement of a piston and the movable contact.

The k-e model included viscous effects that were considered for the sliding wall; consequently, there was a boundary layer. No slip was the default boundary condition to model solid walls. A no-slip wall was a wall where the fluid velocity relative to the

wall velocity was zero. For a stationary wall, it meant that $\mathbf{u} = 0$. The constraint could be mathematically formulated for the problem with a moving wall:

$$\mathbf{u}_{rel} \cdot \boldsymbol{n} = 0 \tag{30}$$

$$\mathbf{u}_{\mathrm{rel}} = \mathbf{u} - \mathbf{u}_{\mathrm{tr}} \tag{31}$$

$$\mathbf{K} - (\mathbf{K} \cdot \mathbf{n})\mathbf{n} = 0 \tag{32}$$

$$\mathbf{K} = \mu \Big(\nabla \mathbf{u}_{rel} + (\nabla \mathbf{u}_{rel})^T \Big) \mathbf{n}$$
(33)

where **n**—boundary normal, with direction outside the region; \mathbf{u}_{rel} is relative velocity; and \mathbf{u}_{tr} is translation velocity.

Turbulence parameters k and ε were subject to homogenous Neumann boundary conditions:

$$\nabla k \cdot \mathbf{n} = 0 \tag{34}$$

$$\nabla \varepsilon \cdot \mathbf{n} = 0 \tag{35}$$

During the calculation, due to the movement of the boundaries of the object, the grid of the computational domain was strongly deformed, which reduced its quality and led to the appearances of discontinuities, inconvergences of equations, etc. In order to eliminate possible errors, it was customary to use the Automatic Remeshing function, which rebuilt the computational mesh when it reached a critically low quality.

4.4. The Proposed Model of Interaction between SF6 Gas Flow and Arc

The model is based on the additional element, which is either the line (axisymmetric) or cylinder (three-dimensional) in the contact gap. The arc column measured temperature is assigned to the line. The temperature is obtained from the experiment of 10 kA symmetrical short-circuit breaking [76].

The temperature change is shown in Figure 9a. The duration of the application of temperature to the moving line was given with the assumption that the arc burned for two power frequency half cycles (i.e., 20 ms). In this case, it was decided to neglect the change in temperature in the vicinity of the zero current. This assumption was very rough because the problem of skipping current zero was extremely important [77–79]. Studies devoted to the analysis of the occurrence of thermal and/or electrical breakdowns due to the passage of zero current gave a complete picture of arcing. However, in this study, the emphasis was on modeling the processes of arcing until the moment of arc extinction, and the processes of restoring electrical strength were not studied.

At the line boundaries (Figure 9b), the temperature of the arc stem was set according to [76] with a current cut of 10 kA. A feature of the model was the fact that the line moved along with the moving contact until the contacts opened (moving mesh). After opening the contacts, the upper point of the line was fixed, and it lengthened after the moving contact. In a three-dimensional picture, this line was represented by a cylindrical arc, on the surface of which the temperature was experimentally set. Thus, gas heating occurs in the contact gap as the contacts are being separated.



Figure 9. (a) Temperature change according to [76]; (b) Arc model with additional line 5 between arcing contacts for adaptive heat release (along the moving contact system): 1—fixed arcing contact; 2—moving arcing contact; 3—auxiliary PTFE nozzle; 4—main PTFE nozzle; A—above-piston volume; B—under-piston volume.

When adding heat sources to the model, the heat balance equation was added to the solution of the above equations for gas dynamics:

$$\rho C_p \frac{\partial T}{\partial t} + \rho C_p \mathbf{u} \cdot \nabla T + \nabla \cdot \mathbf{q} = Q + Q_p + Q_{vd}, \tag{36}$$

(b)

where C_p is the specific heat capacity at constant pressure, $J/(kg \cdot K)$; **q**—heat flow due to thermal conductivity, W/m^2 ; Q—heat source other than viscous dissipation, W/m^3 ; Q_p is the work performed by changing the pressure and is the result of heating during adiabatic compression, as well as some thermoacoustic effects, W/m^3 ; Q_{vd} is viscous dissipation in liquid, W/m^3 .

$$Q_p = \alpha_p T \left(\frac{\partial p_A}{\partial t} + \boldsymbol{u} \cdot \nabla p_A \right), \tag{37}$$

where α_p is the coefficient of thermal expansion, 1/K:

$$\alpha_p = -\frac{1}{\rho} \left(\frac{\partial \rho}{\partial T} \right)_p \tag{38}$$

For heat transport turbulence consideration, the Kays–Crawford model was used (the default in COMSOL Multiphysics). The viscous dissipation, in this case:

$$Q_{\rm vd} = \tau : \nabla \mathbf{u} + Q_{turb} \tag{39}$$

where τ is the viscous stress tensor, *Pa*.

(a)

Heat flux by conduction (RANS turbulence model):

$$q = -(k + k_T)\nabla T \tag{40}$$

with the turbulent thermal conductivity defined as:

$$k_T = \frac{\mu_T C_p}{P r_T} \tag{41}$$

where Pr_T —Prandtl number, according to [80].

In this case, the boundary conditions for the first kind of Dirichlet were written in the form of Equations (41) and (42) on the boundary where there was no heat flux across the boundary (in general form):

-1

$$\mathbf{n} \cdot \mathbf{q} = 0 \tag{42}$$

However, taking the wall treatment into account, the boundary condition is rewritten as:

$$-\mathbf{n} \cdot \mathbf{q} = \mathbf{q}_{wf} = \rho C_p \mathbf{u}_{\tau} \frac{T_w - T}{T^+}$$
(43)

where \mathbf{q}_{wf} —the heat flux between the fluid with temperature *T* and a wall with temperature T_w ; C_p —is the fluid heat capacity; \mathbf{u}_{τ} —is the friction velocity given by wall treatment (Equations (29)–(32)); T_w —wall temperature (on the boundaries which are "temperature source"); and T^+ – is the dimensionless temperature and is given by [81].

The temperature on the boundaries that represent the experimental arc temperature profile (according to [76]):

$$T = T_0 \tag{44}$$

4.5. Calculation Results

Table 6 presents the results of calculation of two turbulent models $(k - \varepsilon \text{ and } k - \omega)$ of the main gas-dynamic parameters: pressure in the under-piston volume p; gas velocity in the nozzle u; mass flow rate G. For convenience, all results are given to contact separation, which are considered in the analytical calculation.

<i>l,</i> mm	<i>t</i> , ms	$p_{k-arepsilon'}$ MPa	$p_{k-\omega}$, MPa	$u_{k-\varepsilon}$, m/s	$u_{k-\omega}$, m/s	$G_{k-\varepsilon}$, kg/s	$G_{k-\omega}$, kg/s
0	0	0.420	0.420	0	0	0	0
6	5.70	0.441	0.441	0	0	0	0
12	8.50	0.462	0.462	0	0	0	0
18	10.80	0.484	0.484	6.0	6.0	0	0
24 30	12.95 15.00	0.507 0.523	0.507 0.523	31.0 59.0	30.0 55.5	0.367 0.699	0.360 0.656
36	16.50	0.533	0.533	83.0	74.5	0.855	0.787
42	17.95	0.552	0.554	105.0	89.5	0.973	0.902
48	19.13	0.572	0.575	121.0	103.0	1.005	0.936
54	20.20	0.590	0.596	121.0	105.0	1.143	1.092
60	21.30	0.607	0.613	123.0	106.0	1.380	1.301
66	22.35	0.626	0.631	105.0	103.0	1.550	1.442
72	23.42	0.644	0.665	119.0	124.0	1.595	1.520
78	24.53	0.659	0.666	126.0	111.0	1.693	1.650
84	25.70	0.673	0.681	119.0	118.0	1.888	1.838
90	27.00	0.682	0.690	145.0	129.0	2.083	2.007
96	28.55	0.676	0.687	98.0	105.0	2.196	2.105
102	30.93	0.645	0.658	105.0	105.0	2.303	2.327
108	33.90	0.590	0.600	83.0	95.0	2.391	2.486
114	37.46	0.511	0.514	65.0	75.0	2.036	2.151
120	45.00	0.383	0.378	20.0	21.0	0.727	0.710

Table 6. Gas dynamic parameters obtained numerically.

Furthermore, the results of the calculations are presented in the form of fields of profiles of speeds, pressures, and temperatures, as the contact system of the arcing device moved when the short circuit currents were turned off. In the profile field, the gas and the corresponding changes in dependent variables (gas velocity, pressure, temperature) are shown in color, and the solid elements of the arc extinguisher involved in the movement (nozzles, contacts, pre-piston areas) are shown in white.

Figure 10 shows the fields of velocity profiles for different positions of the moving part without taking into account the arc (turbulence model $k - \varepsilon$). It can be seen that the gas velocity inside the moving contact cavity exceeded the gas velocity in the nozzle at some moments. However, even in this case, the main volume of gas flowed out through a large nozzle because the cross-sectional area of the nozzle was greater than the cross-sectional area inside the moving contact.



Figure 10. Velocity field without taking arc into account for turbulent model $k - \varepsilon$ (on the left—the contact separation and time).

Figure 11 shows the change in gas pressure at different positions of the moving part without taking into account the arc (turbulence model $k - \varepsilon$). An increase in the pressure in the under-piston and above-piston volumes could be noticed (assumption: it was assumed that there is no valve between them). With a real shutdown of the short-circuit current, the valve between the areas would begin to close, and the arc extinguishing was mainly carried out by blowing from the above-piston volume. Furthermore, the figure basically shows that the pressure in the contact gap was pumped after the opening of the large nozzle (contact separation is 90 mm).



Figure 11. Pressure field without taking arc into account for turbulent model $k - \varepsilon$ (on the left—the contact separation and time).

Figure 12 shows the change in pressure in the under-piston volume for both analytical and the numerical calculations, taking into account the turbulent models $k - \varepsilon$ and $k - \omega$, respectively. According to Figure 12, it can be seen that the pressure in the underpiston volume changed smoothly—almost linearly almost until the very end of the contact separation. This happened due to the neglect of both turbulent SF6 gas flows and the impact of the shutdown arc on this flow. According to the results obtained numerically, it is seen that the pressure peak occurred at the moment of the deceleration of the contact separation, which was t = 27 ms. The decrease in pressure could be explained by the fact that the gas compression rate slowed down, whereas the mass flow continued to increase.



Figure 12. Pressure change in the under-piston volume obtained analytically (according to [15,18]) and numerically for turbulent models $k - \varepsilon$ and $k - \omega$.

Figure 13 shows the gas mass flow change through the section of the under-piston volume. It can be seen that the flow curve was tied to the cross section of the regions that

the SF6 gas passed on the way to the contact gap, as in the analytical calculation. The differences were caused by less linear transitions with an increase in the cross-sectional area of the SF6 gas movement in the direction of the contact system due to turbulence. This meant that the analytical method gave only a superficial understanding of gas dynamics during the CB breaking or making. Considering the turbulent flows (without arc affect) and turbulent heat transport (with arc affect), even with low relative tolerance, gave a different nature of the interrupter's switching process. The proposed numerical model might have been used for the influence estimation of the thermal effect of the switching arc on the nozzles, as well as for the estimation of both the thermal and electric breakdowns.



Figure 13. SF6 gas mass flow through cross section of the under-piston volume for analytical (according to [15,18]) and numerical calculations.

Figure 14 shows the gas velocity change in the narrowest part of the nozzle (confuser) versus time and contact separation, respectively.



Figure 14. Velocity changes in the smallest cross-section of the nozzle depending on the time and contact separation.

Up to a certain point in time ($t_{crit} = 19.13$ ms), the gas velocity increased, after which some constancy could be observed. Therefore, we could assume that the speed had reached its critical value. This meant that the time t_{crit} could be considered a critical point, and the time interval up to this point was a subcritical gas outflow regime. The interval after the critical point and before the moment of braking was a supercritical regime of gas outflow. Velocity fluctuations occurred due to additional outflow of gas into the area of the opened section inside the moving contact.

Figure 15 shows the gas velocity field, taking into account the setting of the arc in the form of a thermal heating source. It can be seen that the gas velocity during opening reached supersonic values (the speed of sound in SF6 gas under normal conditions was 130 - 135 m/s and increased with increasing pressure of the medium). For the process of arc extinguishing, this was a favorable factor because there was an intensification in the cooling of the plasma channel of the cut-off arc. However, one could also notice that a local area was formed along the stationary arcing contact, where the velocities prevailed. This fact was due to the assumption made in the calculation model—the main part of the arc-extinguishing movable contact was cut out. Thus, it was considered that the main blast was directed through a large nozzle (disconnection of high short-circuit currents).



Figure 15. Velocity field taking arc into account for turbulent model $k - \varepsilon$ (on the left—the contact separation and time).

Figure 16 shows the gas pressure field, taking into account the setting of the arc in the form of a thermal heating source. When such a short-circuit current breaking took place, the pressure increased by about 2.0–2.5 times from the nominal value (0.5 MPa) and reached approximately 1.0–1.5 MPa for the self-blast-type interrupters [17].

Figure 17 shows a picture of the temperature profile in the arc quencher during shutdown. In contrast to the arc-free mode, there was a clear change in the SF6 flow in the contact gap due to the occurrence of flow stagnation areas, where the SF6 gas flow, passing through a small nozzle, met the SF6 gas flow going through a large nozzle. However, their directions were opposite to each other. The local heating of the area along the fixed arcing contact was due to the assumption made above.



Figure 16. Pressure field taking arc into account for turbulent model $k - \varepsilon$ (on the left—the contact separation and time).



Figure 17. Temperature field (on the left—the contact separation and time).

Due to the consideration of the arc action, the pressure in the contact gap changed and has a less decreasing character, in contrast to the arc-free mode.

5. Conclusions

In the developed model described above, the temperature characteristics of the arc shaft were obtained from the experiment of switching off the symmetrical short-circuit current of 10 kA [76]. The proposed model of the switching arc was implemented in numerical software using the finite element method based on a moving-mesh technique (ALE method). The ALE method allowed us to take the contact separation curve of the CB into consideration and to make the developed model adaptive. In order to make a comparison with experimental data on breaking higher short-circuit currents, the temperature change was adapted according to [73] for an arcing time of 20 ms (two half cycles of power

frequency). The accuracy of the developed model was determined in comparison with experiments on breaking the symmetrical current of 25 kA by a self-blast interrupter from the study [82]. The comparison was carried out by the means of calculating the coefficient of determination R^2 by changing the pressure in the under-piston volume (Figure 18). Due to the consideration of the arc effect, the pressure in the contact gap changed and had a less decreasing character, in contrast to the arc-free case.



Figure 18. Pressure change in under-piston volume (comparison with [82]).

From Figure 18, it can be seen that the developed model in the form of a temperature heating source had a good quality in comparison with the experiment [82], with a determination coefficient $R^2 = 0.997$.

According to the results of the calculations, it is advisable to note the assumptions and critical parameters of the model:

1. The absence of a valve between the under-piston and above-piston volumes, which clearly affects the velocity profile of the SF6 flow moving into the contact gap.

2. Cut out the main part of the arcing moving contact, leading to a one-way blow (through a large nozzle). Therefore, the model was suitable for breaking large short-circuit currents.

3. The construction of the computational grid was performed by the user (user controlled) and not by the built-in tools of the numerical software complex (physics controlled). As a result, the computational mesh on the boundary layers had the same quality as in the main computational domain. In other words, the calculation of gas dynamics was carried out quite roughly.

4. The buoyancy forces were not taken into account in the calculations of gas dynamics.

5. The assumption of Mach numbers <0.3 was used. However, in the peak phases of the arcing, the SF6 flow velocity exceeded the supersonic flow. Thus, it was more correct to use models that were used for the sonic and supersonic flows' descriptions.

6. To increase the convergence, the accuracy parameter of the moving grid (relative tolerance) was taken as equal to 0.1, which made it possible to consider the physical processes only in the first approximation.

7. The contact separation curve was taken from the arc-free case ("no-load") and was considered unchanged even when the arc was taken into account. However, when the arc was extinguished, the contact separation curve "collapses" [83], which clearly affected the entire process of the gas dynamics, including the gas outflow regime.

8. Despite the fact that the radiation effect was significant within such high temperatures of the switching arc column (>13,000 K), the assumption of the neglecting this effect was used.

As prospects for further research directions, in addition to considering the above assumptions, the following can be noted:

1. Taking all the physical processes of interaction the SF6 gas flow and switching arc (creation of magnetohydrodynamic, hydrokinetic, and kinetic models that make it possible to trace the movement of the conductive medium—the arc plasma in SF6 gas), with the aim of their possible syntheses and the creation of universal approaches to the simulation of the switching arc.

2. Adapt the developed model for "near zero" arc extinguishing processes, taking into account the processes of restoring the electrical strength of the contact gap, the residual conductivity of the arc, and assessing the likelihood of thermal and electrical breakdowns, etc.

3. An improvement of the methods for calculating the equations of gas dynamics in order to minimize computational costs.

4. Analyze the occurrence of computational instability in the calculations of gas dynamics and develop prerequisites for combating this problem without losing the accuracy of the simulation.

Author Contributions: Conceptualization, V.V.P., A.I.K. and Y.V.P.; methodology, A.I.K. and V.V.P.; software, V.V.P. and Y.V.P.; validation, A.I.K. and V.V.P.; writing—original draft preparation, A.I.K., V.V.P. and Y.V.P.; writing—review and editing, A.I.K. and V.V.P.; visualization, V.V.P.; supervision, A.I.K. All authors have read and agreed to the published version of the manuscript.

Funding: The research was carried out within the state assignment with the financial support of the Ministry of Science and Higher Education of the Russian Federation (subject No. FEUZ-2022-0030 Development of an intelligent multi-agent system for modeling deeply integrated technological systems in the power industry).

Conflicts of Interest: The authors declare no conflict of interest.

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